

TPS: From Arc-jet to Flight

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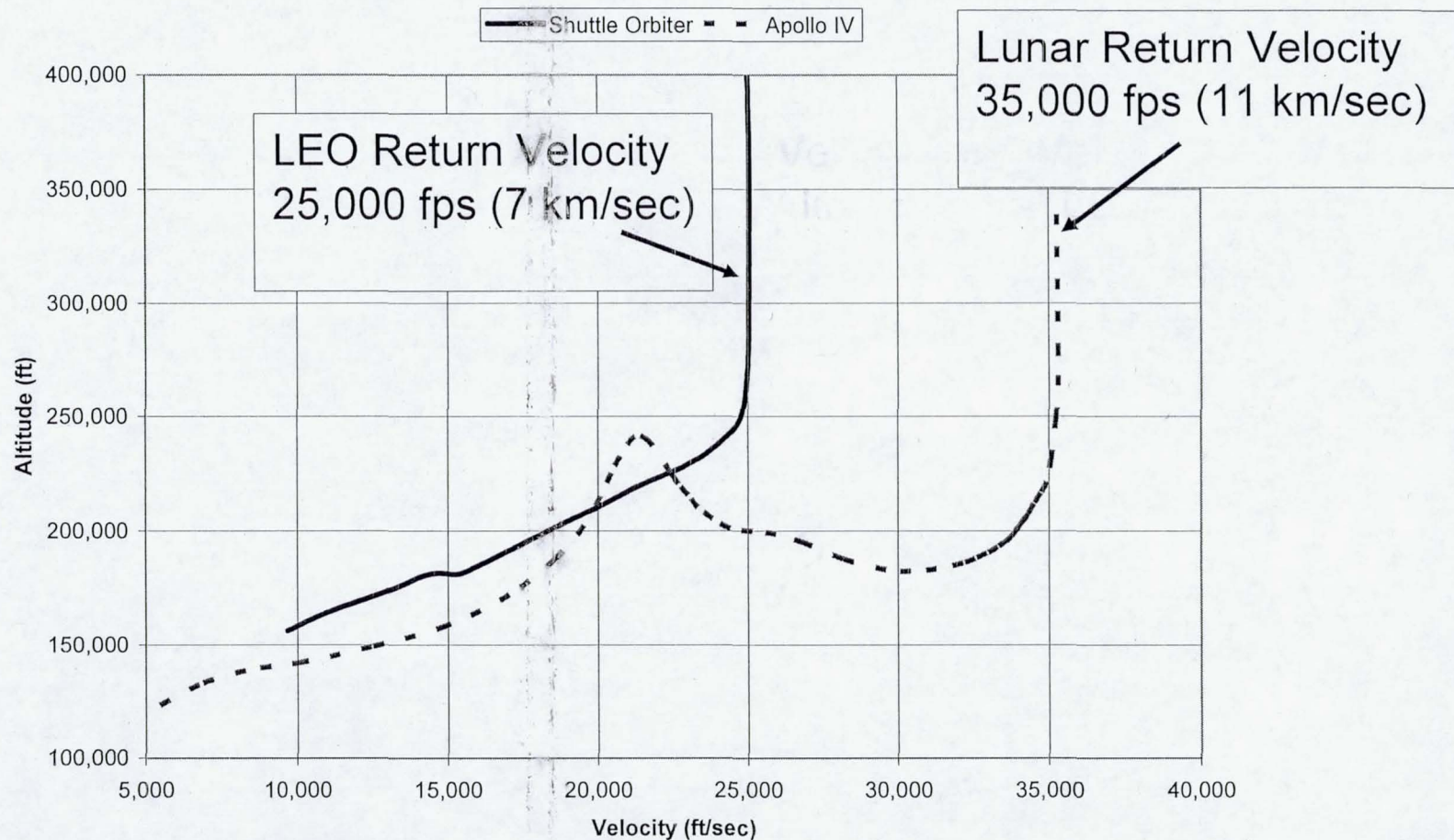
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Background

- All space vehicles that reenter Earth's atmosphere from either LEO or from Lunar/Mars missions require thermal protection system (TPS) materials.
- TPS material development and verification require ground test facilities that simulate reentry aerothermodynamic environments.

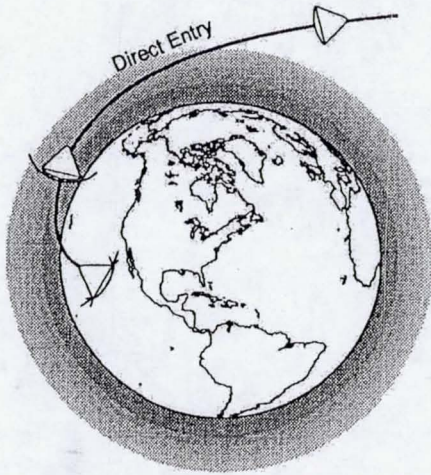
Reentry Trajectory Drives TPS Material Requirements



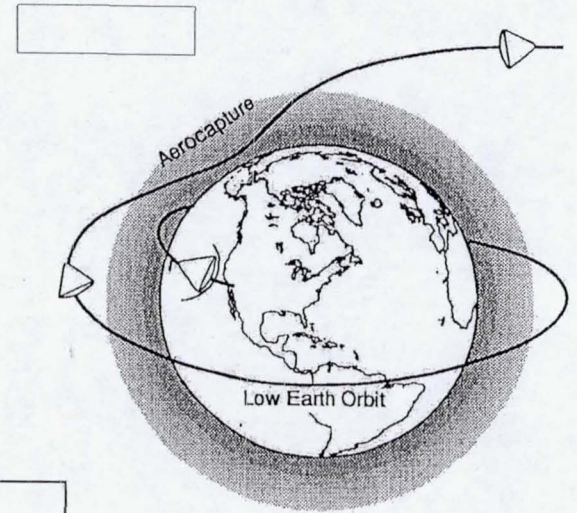
Lunar/Mars Return Aerothermodynamic Environments

- High Enthalpy Gas
 - Lunar/Mars Return: Air @ 20,000+ Btu/lb_m (46 MJ/kg)
- High Convective Heating
 - Lunar/Mars Return: 200+ Btu/ft²-sec (227+ W/cm²)
- High Gas Cap Radiation Heating
 - Lunar/Mars Return: 200+ Btu/ft²-sec (227+ W/cm²)
- High Stagnation Pressures
 - 0.5 atmospheres
- Aerodynamic Shear Stress

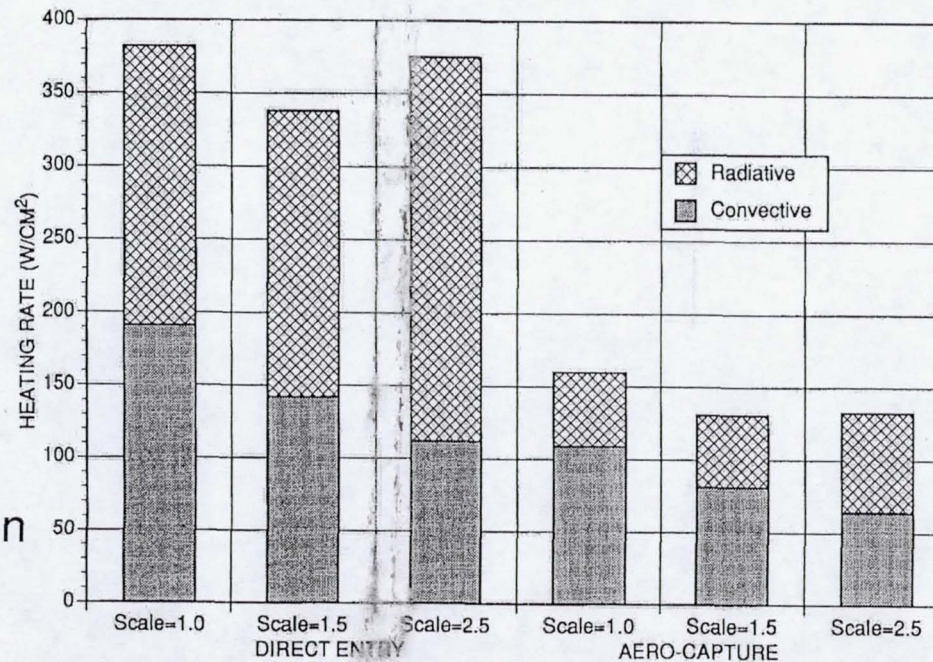
Convective and Radiation Heat Flux



Gas Cap Radiation Heating
Comparable Level as Convective



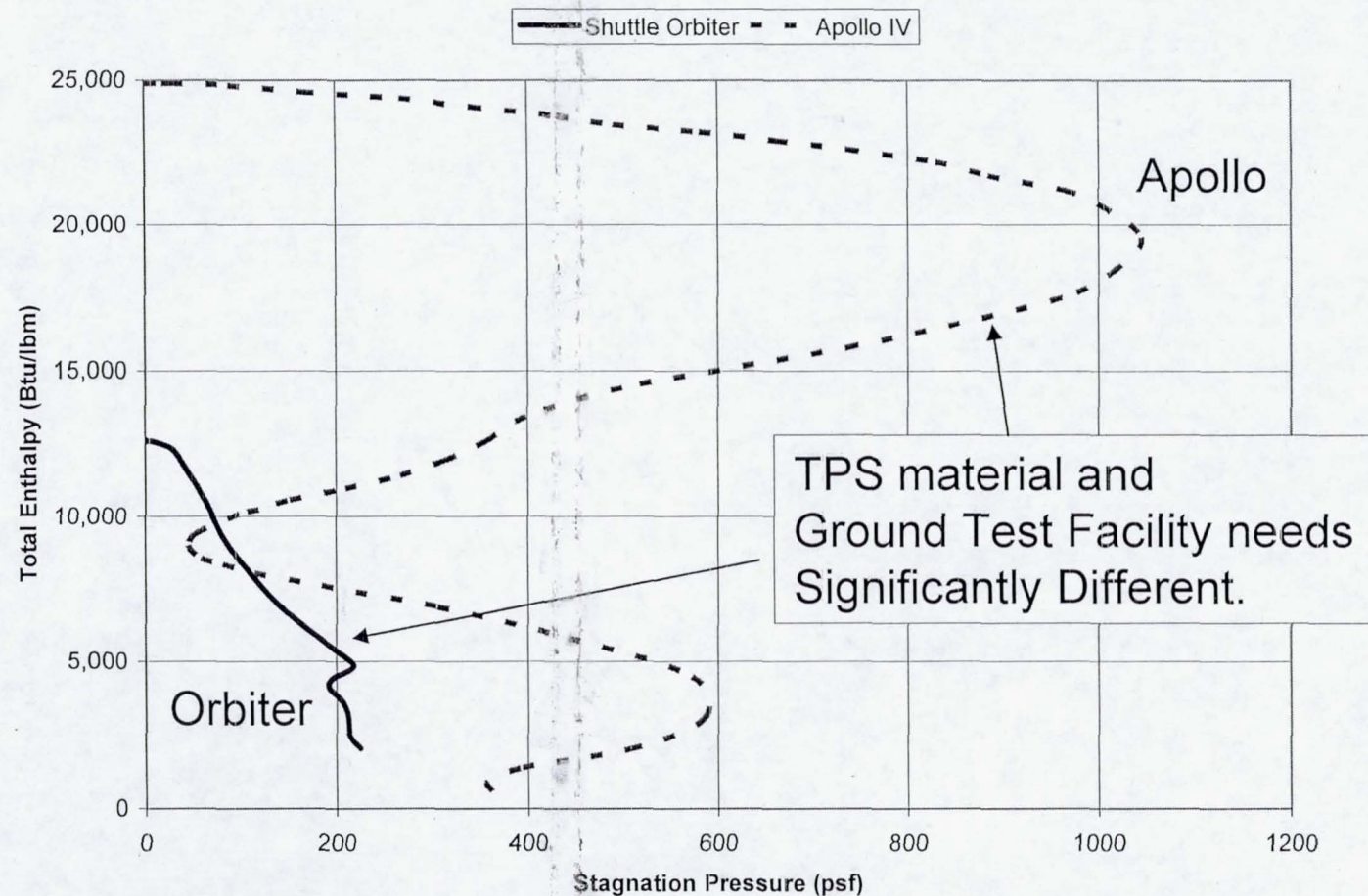
PEAK HEATING RATES TO SCALE APOLLO VEHICLES



Increasing Radius
increases Radiation
and decreases
Convection.

Scaling from Ground Test to Flight

Total Enthalpy and Stagnation Pressure

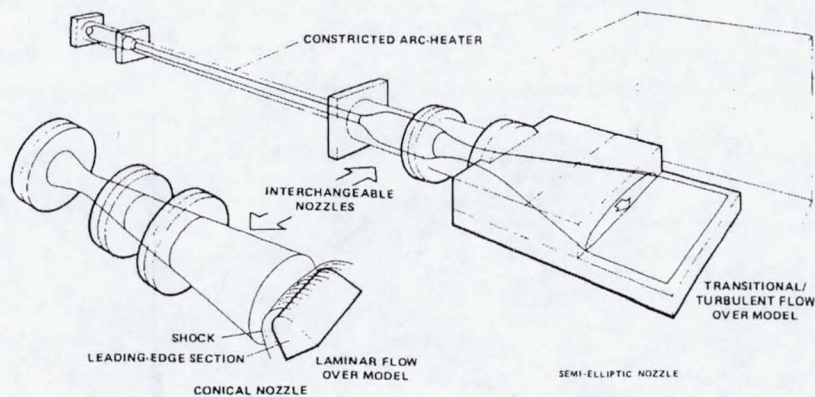


Ground Test Facility Needs

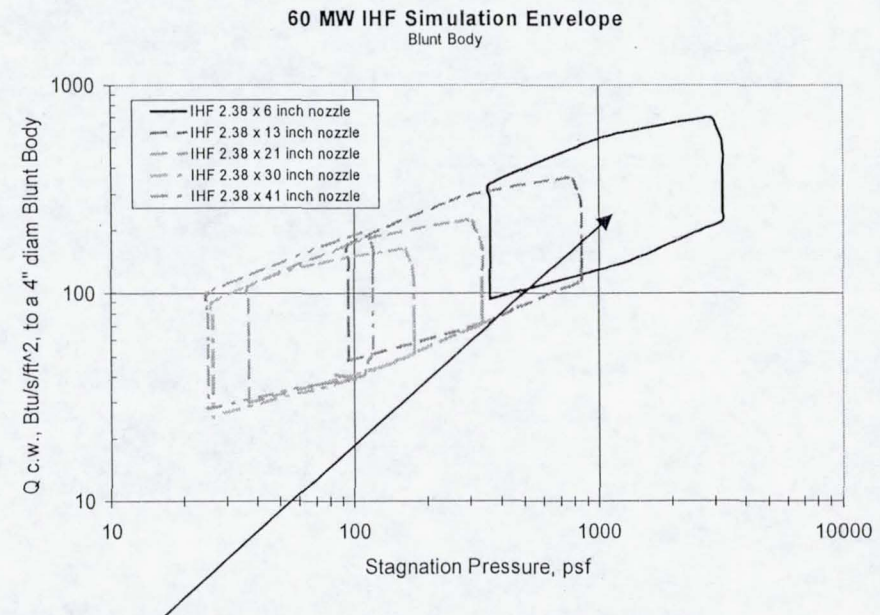
- High Enthalpy Gas Facilities
 - Both air and CO₂
 - Large range of stagnation pressures
- Combined Convective and Radiative Heating

NASA Ames Arc-jet Facility

60MegaWatt Interaction Heating Facility



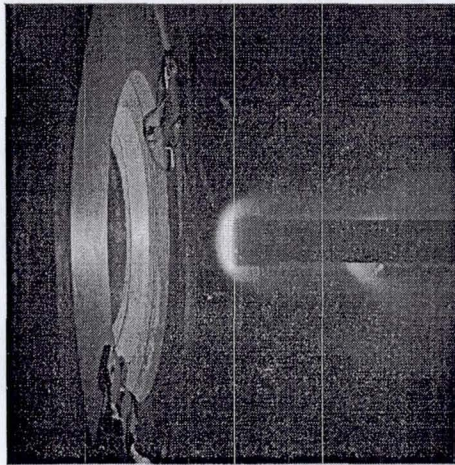
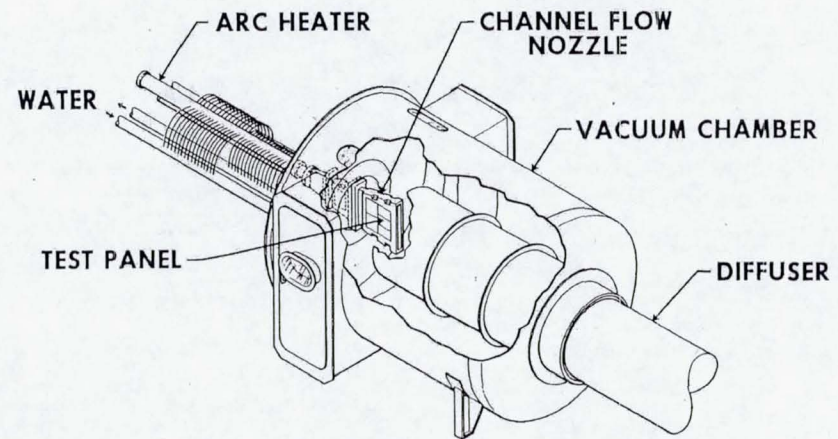
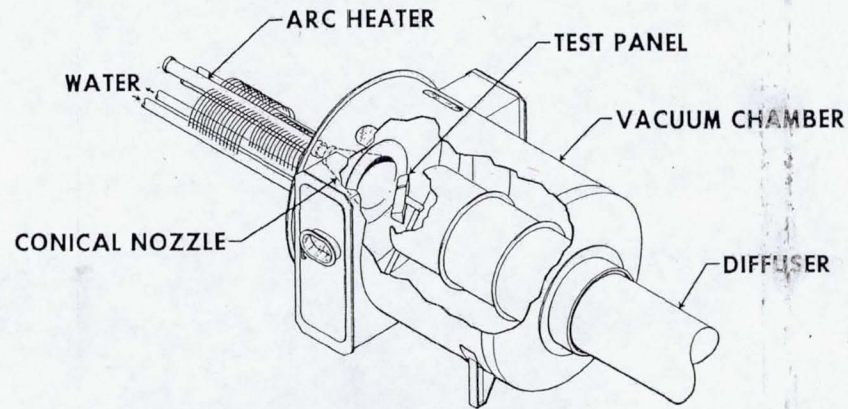
Test Gases: Air and Nitrogen



High Convective Heating and High Stagnation Pressures
can be achieved on small test specimens.

NASA JSC Arc-jet Facilities

Orbiter TPS Certified Facility



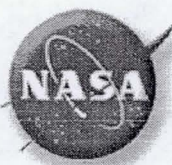
Test Gases: Simulated Air; Nitrogen

Stagnation Heat Fluxes:

Stagnation Pressures:

Concluding Remarks

- Future Lunar/Mars missions require ground test facilities for TPS material development and flight certification.
- Existing arc-jet facility capabilities need to be updated and expanded.
 - Combined convective and radiation heating.
 - CO₂ test capability for Mars entry.



TRANSFORMATIONAL SYSTEMS CONCEPTS & TECHNOLOGIES FOR FUTURE SPACE MISSIONS

A Technical Workshop

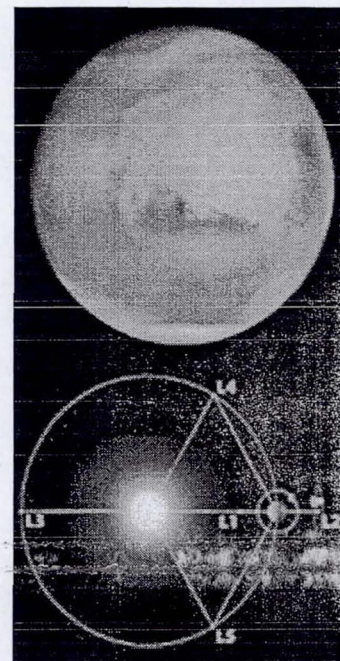
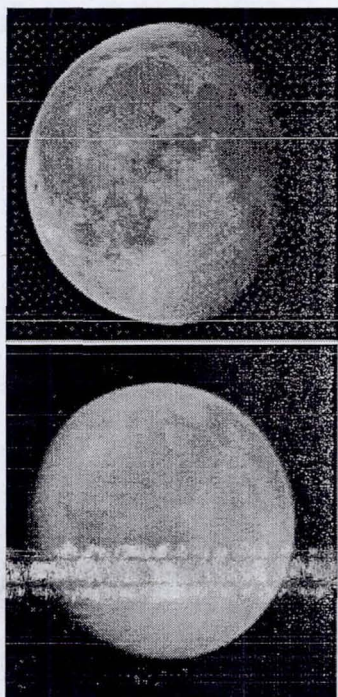
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Hosted Jointly by
The Lyndon B. Johnson Space Center
The Langley Research Center

March 1st thru 3rd, 2004

Hilton Hotel and Conference Center
The University of Houston
4800 Calhoun Rd.
Houston, Texas 77204



OVERVIEW

Monday, 1 March 2004, in the Plaza Room

| <u>TIME</u> | <u>EVENT</u> |
|------------------|---|
| 7:45 – 8:30 AM | Registration and Refreshments |
| 8:30 – 8:45 AM | Welcome |
| 8:45 – 9:30 AM | Overview of NASA Vision for Exploration (TSD) <i>John Mankins, NASA HQ</i> |
| 9:30 – 10:30 AM | Materials Needs for Exploration <i>Mr. John Connell, NASA Langley Research Ctr</i> |
| 10:30 – 10:45 AM | Break |
| 10:45 – 11:45 AM | Revolutionary Materials <i>Dr. Richard Smalley, Nobel Laureate, Rice University: Carbon Nanotubes and Space Applications.</i> Presentation and Discussion |
| 11:45 – 12:45 PM | Lunch |
| 12:45 – 1:45 PM | Carbon Nanotube Woven Garment <i>Mr. Ray H. Baughman</i> <i>Robert A. Welsh Professor of Chemistry and NanoTechnology, University of Texas-Dallas</i> Catalytic Presentations & Discussions |
| 1:45 – 3:45 PM | Break |
| 3:45 – 4:00 PM | Break |
| 4:00 – 5:00 PM | Catalytic Sub Session Reports and Discussion |